

# **Trailing Edge Flaps on Multi-MW Turbine Blades**

**European Bladed User Conference**

**10. November 2016, Hamburg  
Germany**

**Sebastian Perez-Becker**

---

## Introduction

## Model Set-up and Load Calculation

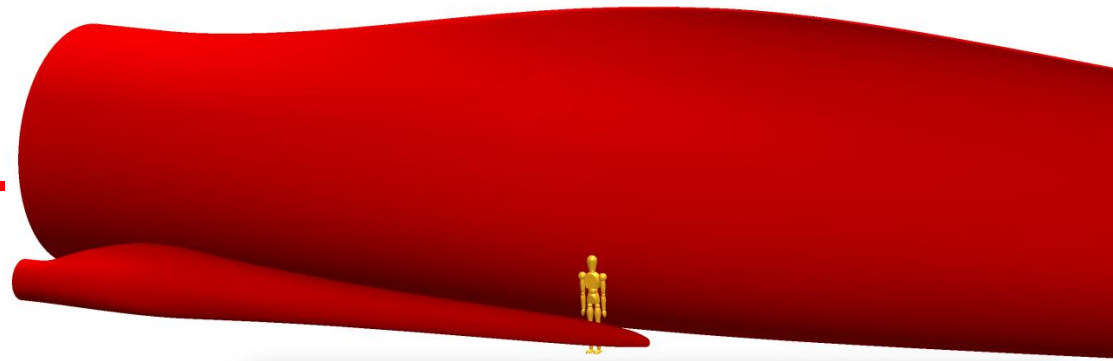
## Results

---

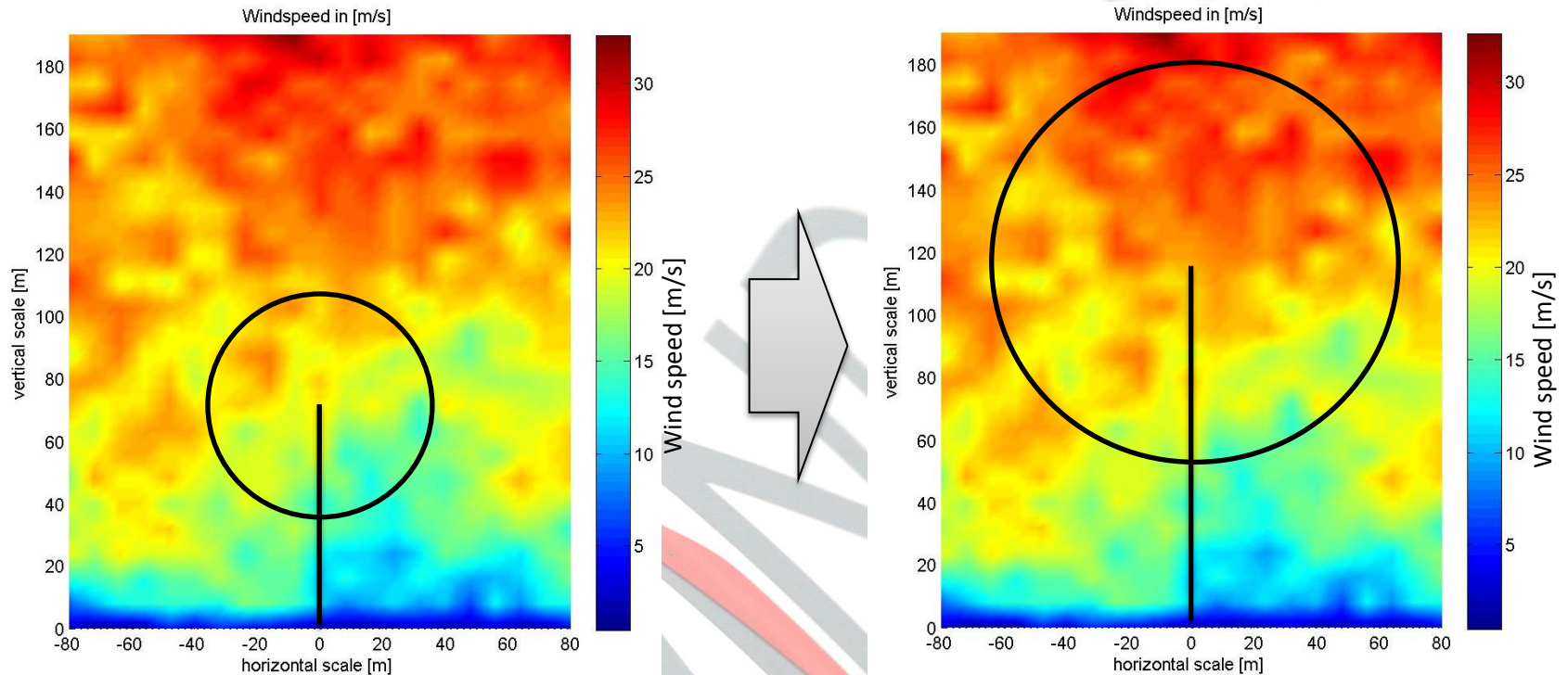
# The Company

---

- Established in Berlin, Germany in 2007
- International team of 27 full time engineers from 6 different countries
- More than 280 projects (100 individual rotor blade designs)
- TechTrans for rotor blade manufacture (India, Korea, China, Canada, USA)
- Wind turbine component design and consulting works



## Why Trailing Edge Flaps?



- Larger blades:
  - Larger spatial turbulence variation / wind shear
  - Slower pitch system
- Flaps: Fast and localized actuation => Limit deflection => Load reduction

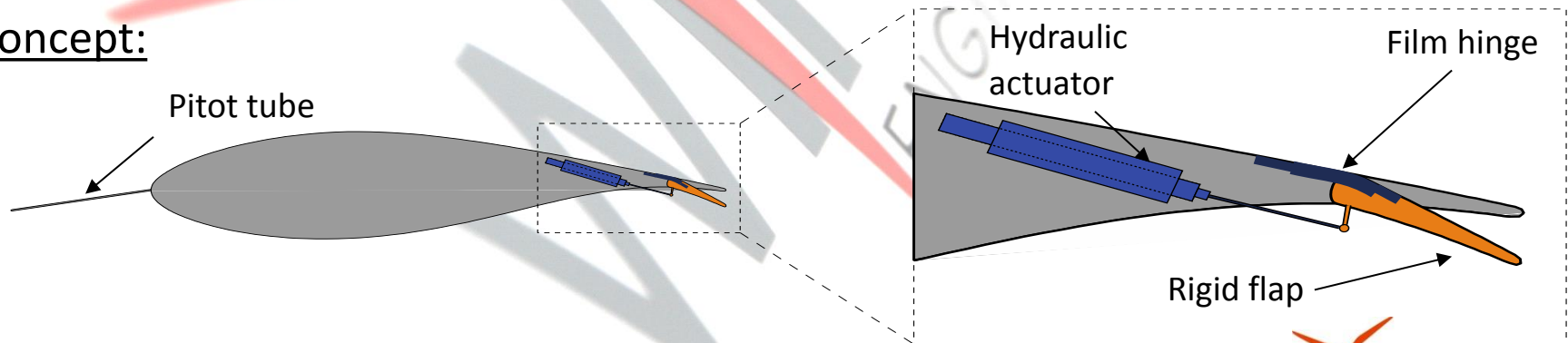
## WIntFlap Project

- Development of an active trailing edge flap
  - Robust
  - Easy maintenance
- Funded by the German Federal Ministry of Economic Affairs and Energy

### Partners:



### Concept:

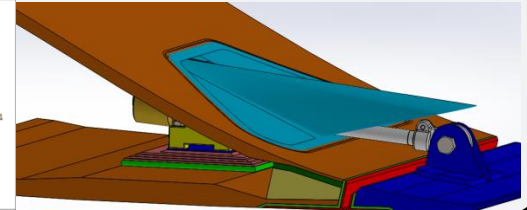
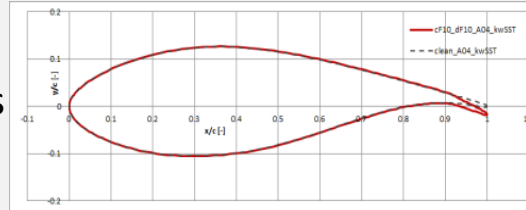




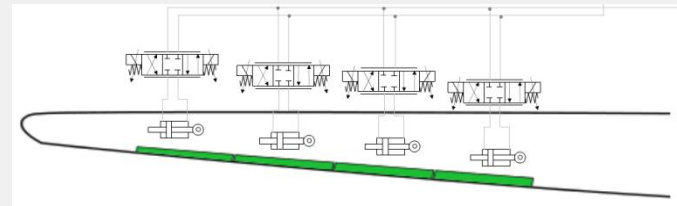
## WIntFlap Project - Tasks



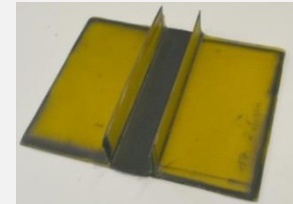
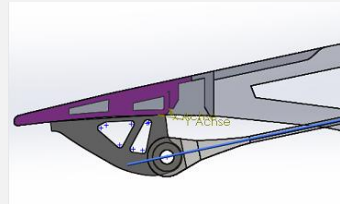
- Aerodynamics
- Controller/Loads
- Blade structure



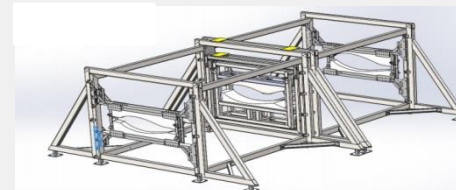
- Actuation
- Lightning protection



- Film hinge
- FI-element



- Flap structure
- Test bench



---

■ Introduction

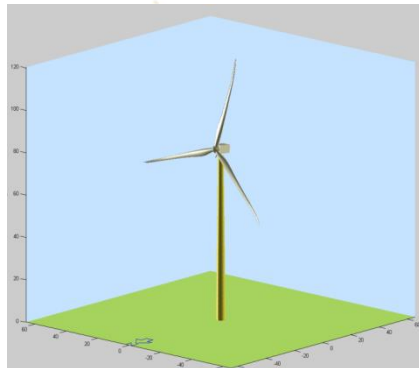
■ **Model Set-up and Load Calculation**

■ Results

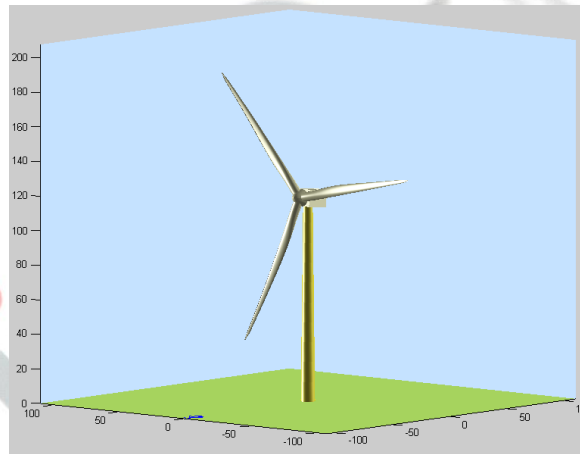
---

## Study Flap Effects: Model Choice

- WIntFlap: Physical design and test of flap for a 1.5 MW rotor blade
- Target blades are in the multi-MW class wind turbines



Results from 1.5MW  
flap design



Translate into 10MW  
turbine model

### DTU 10 MW RWT

Wind class	IEC 1A
Rated power	10 MW
Cut in, Cut out wind speed	4 m/s, 25 m/s
Rated wind speed	11.4 m/s
Rated rotor speed	9.6 rpm
Rotor diameter	178.3 m
Hub height	119 m

Adapt turbine to  
include flaps:

Aerodynamics

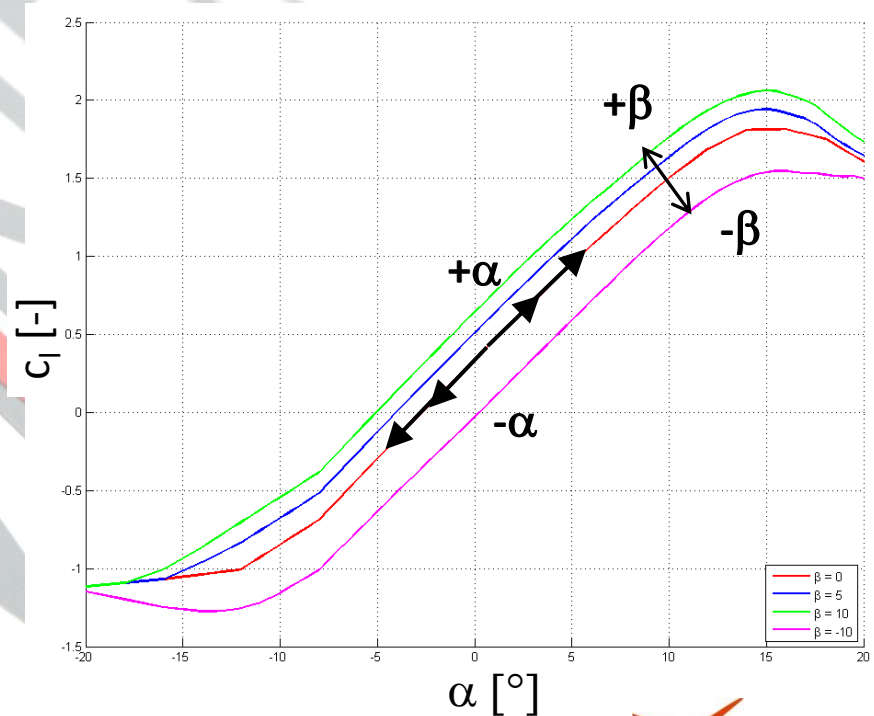
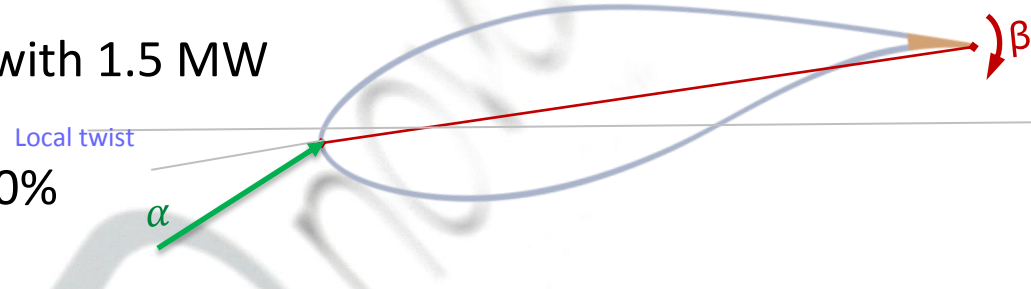
Blade  
structure

Controller



## 1. Aerodynamics

- Same local flap configuration as with 1.5 MW blade
  - Flap chord-wise extension: 10%
  - Flap deflection limit:  $\pm 10^\circ$
- Airfoil: FFA-W3-241
- Effect of flap deflection on airfoil characteristics via 2D-CFD calculations



## 1. Aerodynamics

- Bladed: flaps modelled as ailerons (set of polars per deployment angle)

**Turbine configuration**

**Turbine and Rotor**

Blade: DTU-10MW-RB-Flap (86.4662 m)

Nominal rotor diameter: 178.332 m

Rotor diameter (coned): 178.162 m

Number of blades: 3

Tower height (Ht): 115.63 m

Hub vertical offset (H): 2.75 m

Total hub height (Ht+H): 118.38 m

Blade set angle: 0 deg

Cone angle (C): -2.5 deg

Tilt angle (T): 5 deg

Overhang (O): 7.1 m

Lateral Offset (L): 0 m

Rotational sense: Clockwise

Rotor position: Upwind

Speed Type: Variable

**Control surfaces: Aileron**

Transmission: Gearbox

Cut-in windspeed: 4 m/s

Cut-out windspeed: 25 m/s

View turbine graphic ...

Apply Reset Mass totals...

Blade Information		Blade Geometry								Mass and Stiffness		Additional Mass/Inertia		
		34	35	36	37	38	39	40 in	40 out	41	42			
Distance along blade	m	78.73691	79.97961	81.13421	82.20538	83.19751	84.11519	84.96301	84.96301	85.74522	86.46616			Add
Distance along blade root Z-axis	m	78.6705	79.9085	81.0585	82.1252	83.113	84.0265	84.8703	84.8703	85.6487	86.366			Delete
Chord	m	2.2324	2.1291	2.0191	1.9033	1.7761	1.626	1.4401	1.4401	1.1836	0.6			Split
Aerodynamic twist	deg	-2.469448	-2.641335	-2.801764	-2.945003	-3.071054	-3.179916	-3.271589	-3.271589	-3.357533	-3.426288			Join
Thickness	%	24.1	24.1	24.1	24.1	24.1	24.1	24.1	24.1	24.1	24.1			Copy
Neutral axis (x)	m	-2.6107	-2.7187	-2.8217	-2.9195	-3.0121	-3.0996	-3.182	-3.182	-3.2592	-3.3315			Paste
Neutral axis (y)	m	0	0	0	0	0	0	0	0	0	0			
Neutral axis, local (x')	%	0	0	0	0	0	0	0	0	0	0			
Neutral axis, local (y')	%	35	35	35	35	35	35	35	35	35	35			
Foil section		6	6	6	6	6	6	6	5	5	5			
Moving/fixd		Moving	Moving	Moving	Moving	Moving	Moving	Moving	Fixed	Fixed	Fixed			
User axis (x)	%	0	0	0	0	0	0	0	0	0	0			
User axis (y)	%	35	35	35	35	35	35	35	35	35	35			

Split station

**Define Aerofoil Sections**

Blade Station: 84.96301m

Section Type: Aileron

Section Number: 6 New

Aerofoil Datasets For Interpolation on Aileron Deployment Angle

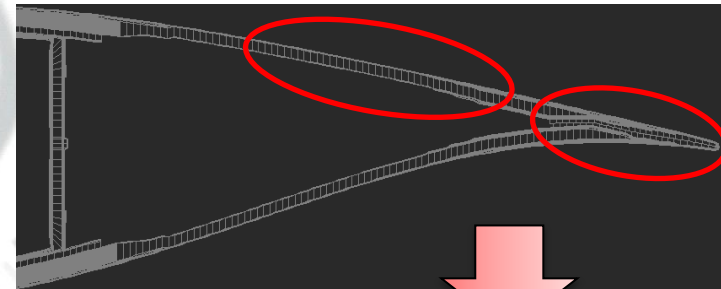
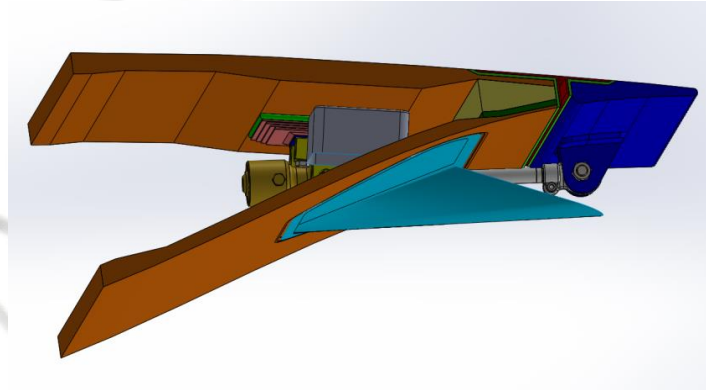
Angle (deg): -10 0 5 10 Add

Aerofoil Dataset: FFA-W' FFA-W' FFA-W' FFA-W' Insert

Delete

## 2. Blade Structure

- 1.5 MW actuation device requires rearrangement of trailing edge structure
  - Trailing edge spar cap has to be modified
- Compensation measure:
  - Reinforcement of main spar cap
  - Reinforcement of trailing edge shell
- Consequence:
  - Mechanical equivalent model affected significantly
  - Higher mass due to reinforced spar cap
- Changes in 1.5 MW equivalent model scaled up for 10 MW model

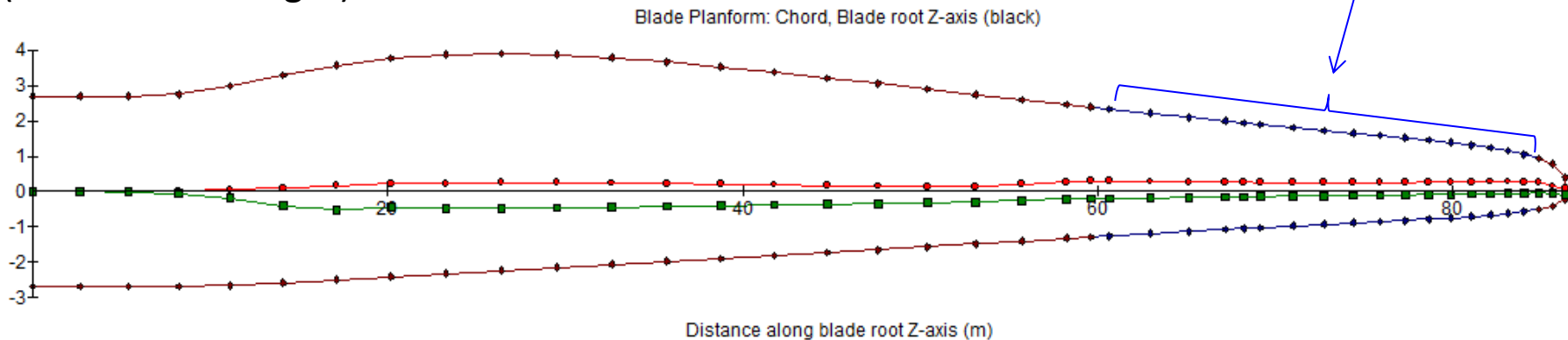


## 2. Blade Structure

Flap position

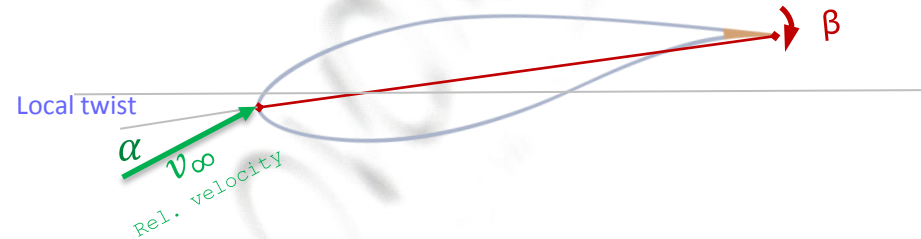
### Flap

Span-wise location: 59.6m – 84.9m  
(~30% blade length)



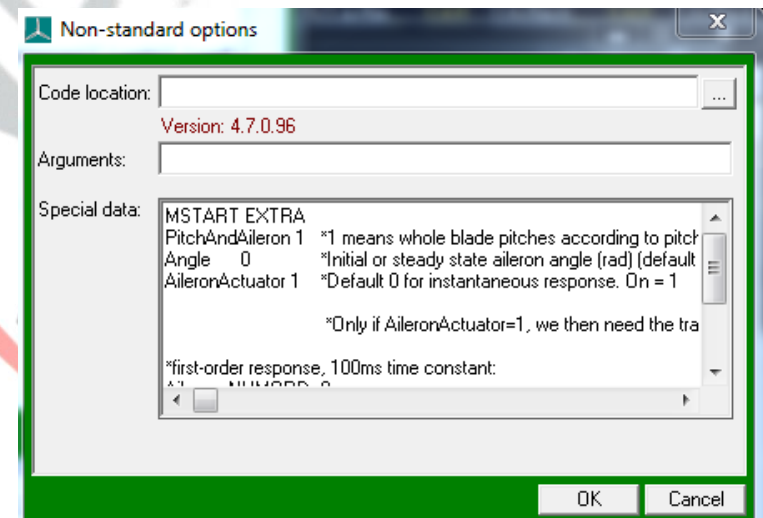
## 3. Controller

- Flap controller independent of pitch/torque controller
- Decentralized concept
  - Act on local aerodynamic changes
  - Taken from Madsen2013<sup>1</sup>
  - Simple feed forward control
- Limit influence on energy capture:
  - Only active at rated power
- Idea: Use flap to keep lift constant
  - Limit blade deflection / turbine loads
- Sensor:
  - Pitot tube: measures  $\alpha$  and  $v_\infty$  in outer blade (ideal sensor in Bladed)



## 3. Controller

- In Bladed:
  - Control of pitch and flap via external controller (using API interface)
- Required non-standard options (Project Info):
  - **PitchAndAileron** (To control pitch and flap)
  - (Aileron initial) **Angle**
  - **AileronActuator** (transfer function)
    - Coefficients





## Load Cases and Variants

- According to GL Guideline for the Certification of Wind Turbines 2010
  - 1.1 Power production with NTM
  - 6.4 Idling with NTM

- Considered fatigue loads (lifetime DEL):

Blade root  $M_{XR}$  (Edgewise)  
Blade root  $M_{YR}$  (Flapwise)

S/N slope:  $m = 10$

Non-rotating hub  $M_{YN}$  (Tilt)  
Non-rotating hub  $M_{ZN}$  (Yaw)

Tower base  $M_{XT}$  (Side - Side)  
Tower base  $M_{YT}$  (Fore - Aft)

S/N slope:  $m = 4$

- Considered variants:

**Original**

- 10 MW w. original blade structure
- No flap
- Reference

**Flap**

- 10 MW w. modified structure incl. flap
- Active flap

---

■ Introduction

■ Model Set-up and Load Calculation

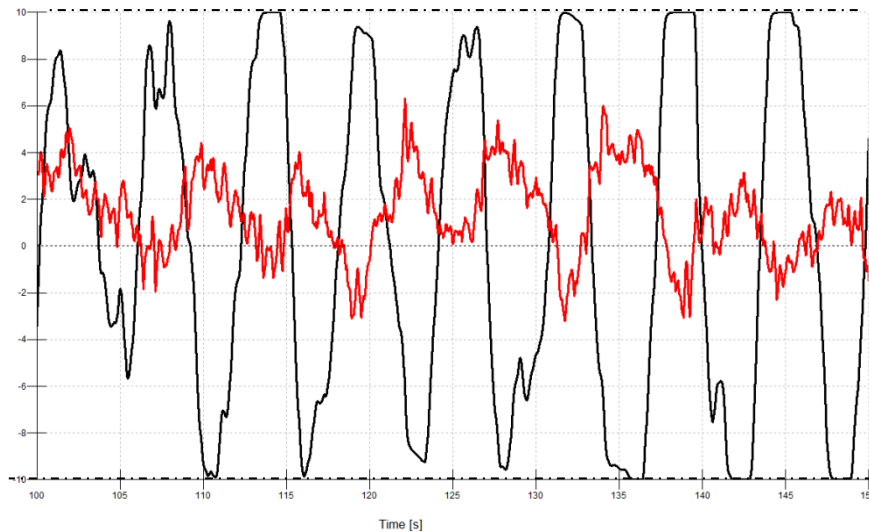
■ **Results**

---

## Time Series

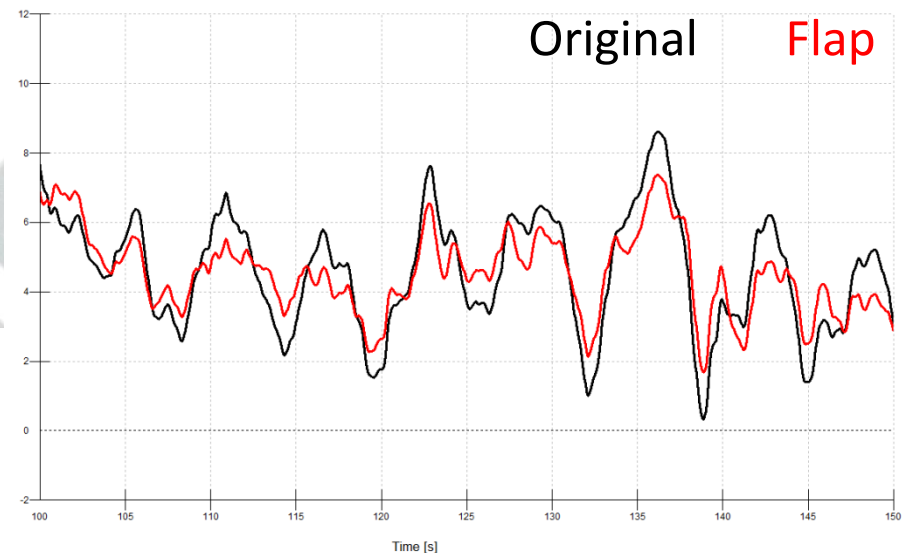
Power production with 15 m/s mean wind speed

Behavior flap control



Flap angle [deg]

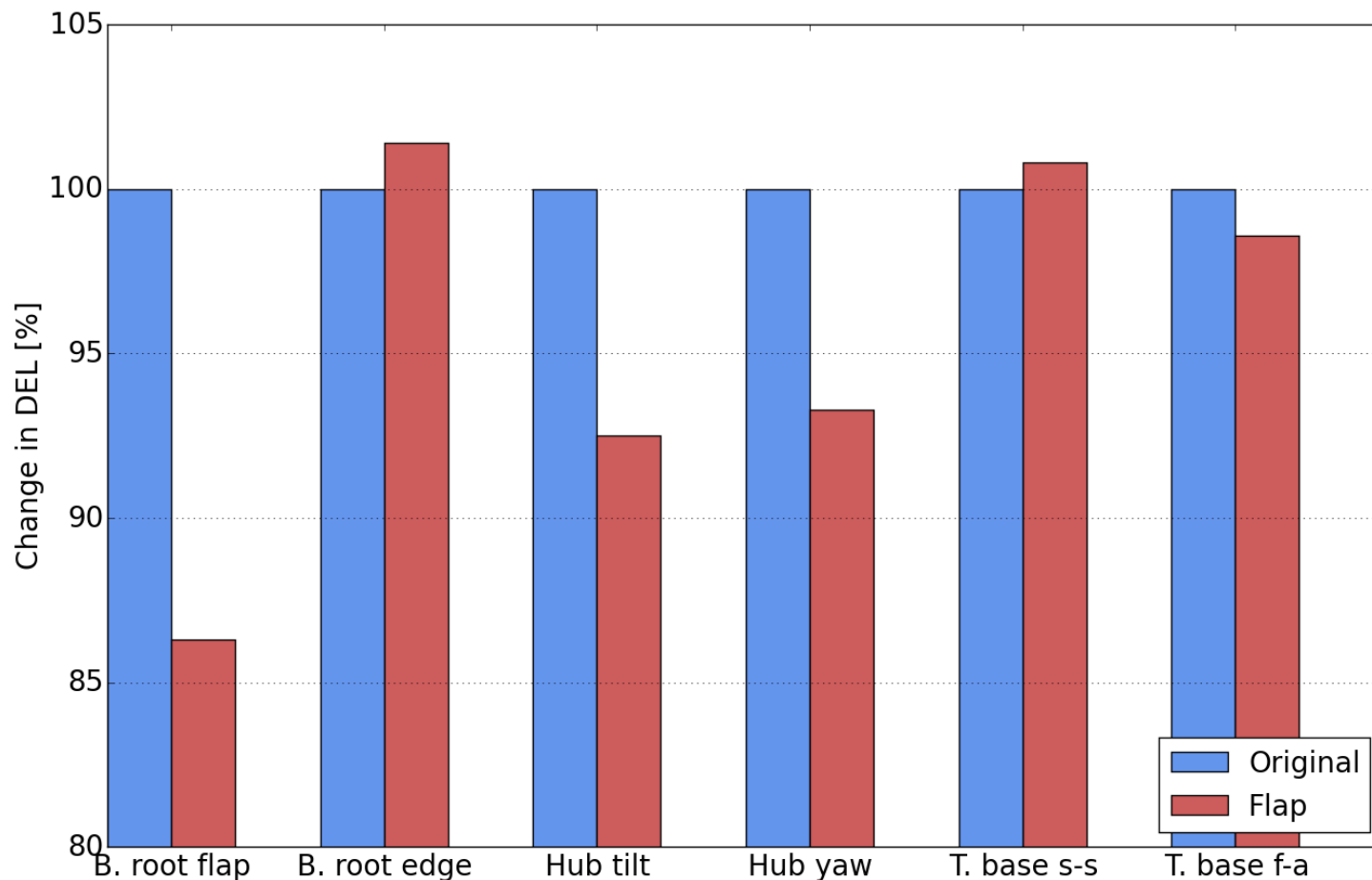
AoA [deg]



Blade tip out-of-plane deflection [m]

## Effect on Fatigue Loads

Comparison of lifetime DELs (GL2010 DLC 1.1 and 6.4)

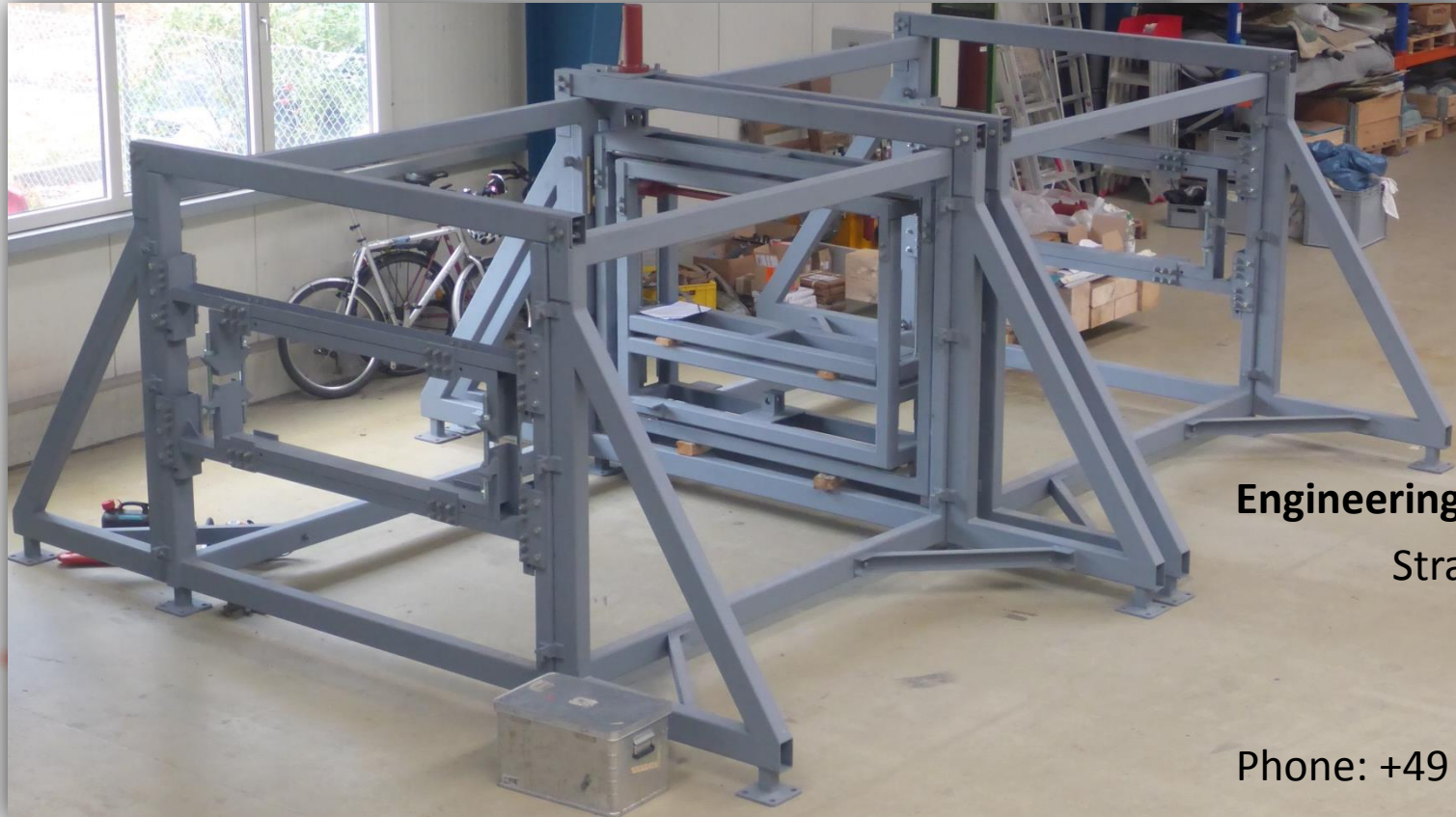


- + Promising fatigue load reduction potential of flaps:
  - Blade root flapwise bending moment
  - Hub tilt/yaw bending moment
- - Blade structural modification increases blade mass / static moment
  - Negative effect on blade root edgewise bending moment
  - Limiting factor on flap size
- Impact on blade design loads?
  - Include extreme loads

- Study effect of other flap control strategies / conventional sensors (i.e. strain gauges)
  - Include extreme loads
- Analyze flap effects on blade design
  - Detailed redesign of existing multi-MW blade to include flaps
  - Complete load calculation (extreme and fatigue loads)
  - Adapt blade structure to new load set



# Thank you for your attention



Source: cp.max

Contact:  
**WINDnovation**  
**Engineering Solutions GmbH**  
Stralauer Platz 33-34  
10243 Berlin  
Germany  
Phone: +49 (0) 30 547195-34  
Fax: +49 (0) 30 547195-39

[www.windnovation.com](http://www.windnovation.com)

E-mail: [sebastian.perez-becker@windnovation.com](mailto:sebastian.perez-becker@windnovation.com)